

# State Space Design: Controllability & Observability

MEM 355 Performance Enhancement of Dynamical Systems

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# Outline

*State space techniques emerged around 1960. They are direct and exploit the efficient computations of linear algebra.*

- State space models, modes & similarity transformations
- Controllability & Observability
- Special forms of state equations
- State Space to Transfer Function

# Similarity Transformations

$$\begin{array}{l} \dot{x} = Ax + Bu \\ y = Cx + Du \end{array} \quad x \in R^n, u \in R^m, y \in R^p \quad \begin{array}{l} \dot{x} = Ax + bu \\ y = cx + du \end{array}$$

Now consider the transformation to new states  $z$ , defined by

$$\begin{array}{l} x = Tz \Leftrightarrow z = T^{-1}x \\ T\dot{z} = ATz + Bu \\ y = CTz + Du \end{array} \Rightarrow \begin{array}{l} \dot{z} = T^{-1}ATz + T^{-1}Bu \\ y = CTz + Du \end{array}$$

so that,

$$\begin{array}{l} \dot{z} = A^*z + B^*u \\ y = C^*z + D^*u' \end{array} \quad A^* = T^{-1}AT, \quad B^* = T^{-1}B, \quad C^* = CT, \quad D^* = D$$

# Diagonal Form

eigen-system

of  $A$ :  $\begin{matrix} \lambda_1 & \lambda_2 & \cdots & \lambda_n \\ h_1 & h_2 & \cdots & h_n \end{matrix}$   $\leftarrow$  eigenvalues  
 $\leftarrow$  independent eigenvectors

$$T \triangleq [h_1 \quad h_2 \quad \cdots \quad h_n]$$

$$\Rightarrow A^* = [h_1 \quad h_2 \quad \cdots \quad h_n]^{-1} A [h_1 \quad h_2 \quad \cdots \quad h_n]$$

$$= \begin{bmatrix} \lambda_1 & 0 & \cdots & 0 \\ 0 & \lambda_2 & \ddots & \vdots \\ \vdots & \ddots & \ddots & 0 \\ 0 & \cdots & 0 & \lambda_n \end{bmatrix}$$

$$\dot{z}_i = \lambda_i z_i + b_i^* u, \quad i = 1, \dots, n$$

A decoupled system  
of  $n$  1<sup>st</sup> order ode's

# Example

Define A

```
>> A=[3 2 1;4 5 6;1 2 3];
```

```
>> [V,D]=eig(A)
```

```
V =
```

```
-0.3482    -0.8581     0.4082  
-0.8704     0.1907    -0.8165  
-0.3482     0.4767     0.4082
```

Compute eigensystem

```
D =
```

```
9.0000         0         0  
0         2.0000         0  
0         0    -0.0000
```

Check similarity trans

```
>> inv(V)*A*V
```

```
ans =
```

```
9.0000    -0.0000    -0.0000  
-0.0000     2.0000    -0.0000  
-0.0000    -0.0000    -0.0000
```

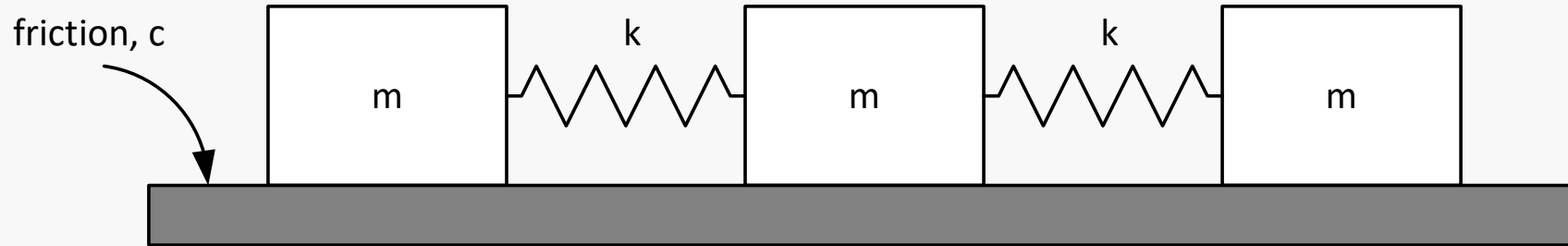
Use linear solve rather  
than inv

```
>> V\A*V
```

```
ans =
```

```
9.0000    -0.0000     0.0000  
-0.0000     2.0000         0  
-0.0000    -0.0000     0.0000
```

# 3 Mass (1)



$$m\ddot{x}_1 = k(x_2 - x_1) - c\dot{x}_1$$

$$m\ddot{x}_2 = -k(x_2 - x_1) + k(x_3 - x_2) - c\dot{x}_2$$

$$m\ddot{x}_3 = -k(x_3 - x_2) - c\dot{x}_3$$

$$\Rightarrow \frac{d}{dt} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ p_1 \\ p_2 \\ p_3 \end{bmatrix} = \begin{bmatrix} 0 & 0 & 0 & 1/m & 0 & 0 \\ 0 & 0 & 0 & 0 & 1/m & 0 \\ 0 & 0 & 0 & 0 & 0 & 1/m \\ -k & k & 0 & c/m & 0 & 0 \\ k & -2k & k & 0 & c/m & 0 \\ 0 & k & -k & 0 & 0 & c/m \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ p_1 \\ p_2 \\ p_3 \end{bmatrix}$$

# 3 Mass (2)

```
> A=[0 0 0 1 0 0;0 0 0 0 1 0;0 0 0 0 0 1;-1 1 0 -1 0 0;1 -2 1 0 -1 0;0 1 -1 0 0 -1];
```

```
> [v,e]=eig(A)
```

```
v =
```

```
Columns 1 through 4
```

```
0.1954 - 0.0589i    0.1954 + 0.0589i    0.4330 - 0.2500i    0.4330 + 0.2500i  
-0.3909 + 0.1179i  -0.3909 - 0.1179i  -0.0000 - 0.0000i  -0.0000 + 0.0000i  
0.1954 - 0.0589i    0.1954 + 0.0589i  -0.4330 + 0.2500i  -0.4330 - 0.2500i  
0.0000 + 0.3536i    0.0000 - 0.3536i  -0.0000 + 0.5000i  -0.0000 - 0.5000i  
-0.0000 - 0.7071i  -0.0000 + 0.7071i   0.0000 - 0.0000i   0.0000 + 0.0000i  
0.0000 + 0.3536i    0.0000 - 0.3536i   0.0000 - 0.5000i   0.0000 + 0.5000i
```

```
Columns 5 through 6
```

```
-0.5774           -0.4082  
-0.5774           -0.4082  
-0.5774           -0.4082  
-0.0000           0.4082  
0.0000            0.4082  
-0.0000           0.4082
```

$$\begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ p_1 \\ p_2 \\ p_3 \end{bmatrix}$$

2 complex conjugate modes (2 eigenvalues/eigenvectors each)  
2 real modes (1 eigenvalue/eigenvector each)

# 3 Mass (3)

```
e =  
Columns 1 through 4  
-0.5000 + 1.6583i      0      0      0  
      0      -0.5000 - 1.6583i      0      0  
      0      0      -0.5000 + 0.8660i      0  
      0      0      0      -0.5000 - 0.8660i  
      0      0      0      0  
Columns 5 through 6  
      0      0  
      0      0  
      0      0  
      0      0  
0.0000      0  
      0      -1.0000
```



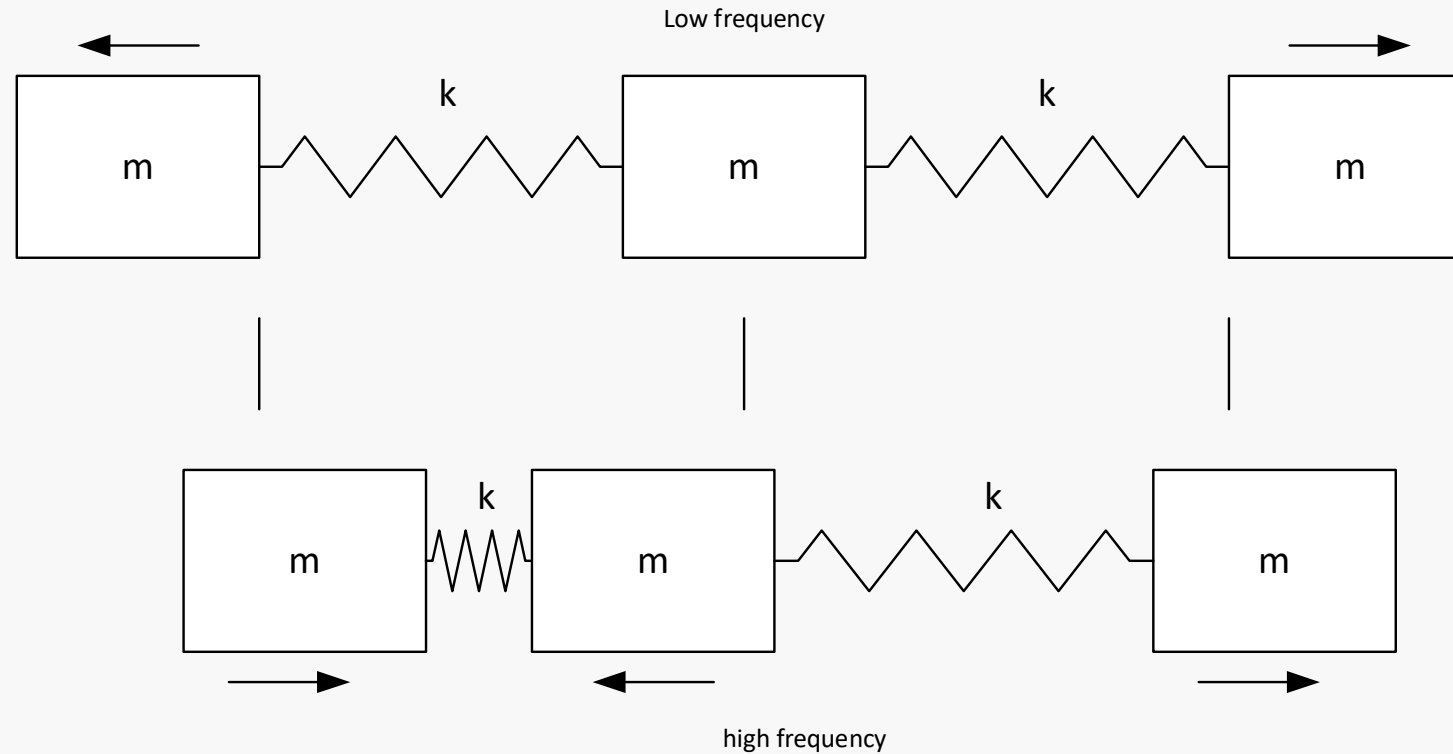
# 3 Mass (4)

Translation

Slow exponential decay

Slow damped oscillation

Fast damped oscillation



# Controllability & Observability

$$\begin{aligned}\dot{x} &= Ax + Bu \\ y &= Cx + Du\end{aligned}$$

**Controllability:** The system is (completely) controllable if there exists a control input  $u(t)$  defined on a finite time interval  $[0, T]$  that steers the system from any initial state  $x_0$  to any final state  $x_1$ .

**Observability:** The system is (completely) observable if the initial state  $x_0$  can be determined from knowledge of the input  $u(t)$  and the measurement of the output  $y(t)$  over a finite time interval  $[0, T]$ .

# Controllability / Observability Tests

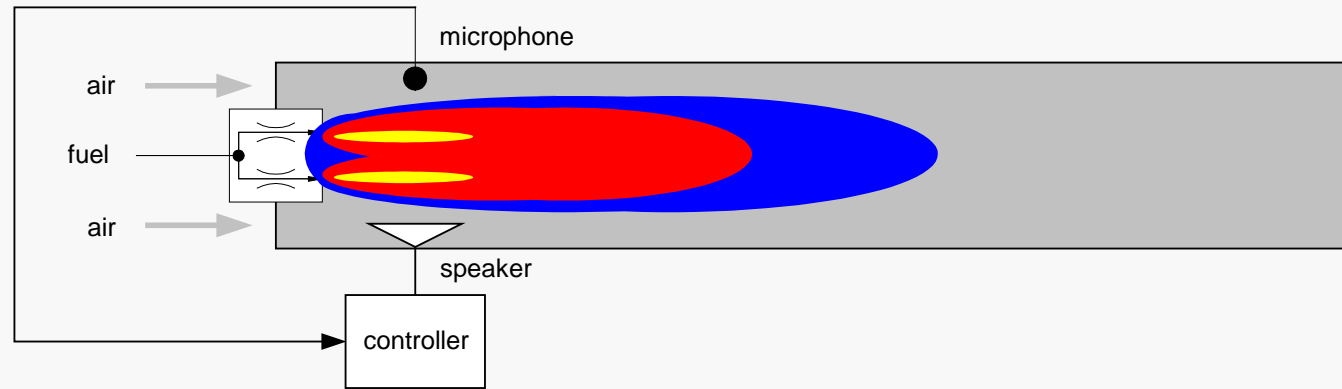
Controllability Matrix:  $\mathcal{C} = \begin{bmatrix} B & AB & \cdots & A^{n-1}B \end{bmatrix}$

Observability Matrix:  $\mathcal{O} = \begin{bmatrix} C \\ CA \\ \vdots \\ CA^{n-1} \end{bmatrix}$

Controllable  $\Leftrightarrow \text{Rank } \mathcal{C} = n$

Observable  $\Leftrightarrow \text{Rank } \mathcal{O} = n$

# Combustion Example



$$G(s) = K \underbrace{\left( \frac{s + z_f}{s + p_f} \right)}_{\text{flame}} \frac{(s^2 + 2\rho_z\omega_zs + \omega_z^2)}{\underbrace{(s^2 - 2\rho_1\omega_1s + \omega_1^2)}_{\text{acoustics}}(s^2 + 2\rho_2\omega_2s + \omega_2^2)}$$

# Combustion Example

```
>> s=tf('s');
zf = 1.500; pf = 1.000; rhoz = -0.045; omegaz = 4.500;
rho1 = 0.5; omegal = 1.0; rho2 = 0.3; omega2 = 3.500;
Gp=((s + zf)*(s^2 + 2*rhoz*omegaz*s + omegaz^2))/((s + pf)*(s^2 -
2*rho1*omegal*s + omegal^2)*(s^2 + 2*rho2*omega2*s + omega2^2));
Gpss=ss(Gp,'min');
[A,B,C,D]=ssdata(Gpss)
A =
    -2.1000    -1.5313    -0.0313    -0.0164    -0.0479
     8.0000         0         0         0         0
         0     4.0000         0         0         0
         0         0     4.0000         0         0
         0         0         0     2.0000         0
B =
    0.5000
     0
     0
     0
     0
C =
     0     0.2500     0.0684     0.3069     0.2373
D =
     0
```

# Combustion Example Cont'd

```
>> Co=ctrb(A,B)
Co =
    0.5000   -1.0500   -3.9200   20.5945    4.7715
         0    4.0000   -8.4000  -31.3600  164.7560
         0     0    16.0000  -33.6000 -125.4400
         0     0     0     64.0000 -134.4000
         0     0     0     0     128.0000

>> rank(Co)
ans =
     5

>> Ob=obsv(A,C)
Ob =
         0    0.2500    0.0684    0.3069    0.2373
    2.0000    0.2737    1.2277    0.4746         0
   -2.0100    1.8481    1.8359   -0.0328   -0.0957
  19.0060   10.4216   -0.0684   -0.1584    0.0962
  43.4599  -29.3767   -1.2277   -0.1195   -0.9095

>> rank(Ob)
ans =
     5
```

# Special Forms

Consider a SISO controllable & observable system

$$\mathcal{C} = [b \quad Ab \quad \cdots \quad A^{n-1}b], \det \mathcal{C} \neq 0 \quad \mathcal{O} = \begin{bmatrix} c \\ cA \\ \vdots \\ cA^{n-1} \end{bmatrix}, \det \mathcal{O} \neq 0$$

$$\mathcal{C}^{-1} \triangleq \begin{bmatrix} q_1 \\ \vdots \\ q_n \end{bmatrix} \quad \mathcal{O}^{-1} \triangleq [p_1 \quad \cdots \quad p_n]$$

We will consider four state transformations defined by

$$T_1 \triangleq \mathcal{C}, T_2 \triangleq \mathcal{O}, T_3 \triangleq \begin{bmatrix} q_n \\ q_n A \\ \vdots \\ q_n A^{n-1} \end{bmatrix}, T_4 \triangleq [p_n \quad Ap_n \quad \cdots \quad A^{n-1}p_n]$$

# Controllability Form for SISO Systems

$$\dot{x} = Ax + bu, \quad y = cx$$

$$\mathcal{C} = [b \quad Ab \quad \cdots \quad A^{n-1}b], \quad \det \mathcal{C} \neq 0$$

$$T = [b \quad Ab \quad \cdots \quad A^{n-1}b]$$

$$T = [A^{n-1}b \quad \cdots \quad Ab \quad b]$$

$$\Downarrow$$

$$\Downarrow$$

$$\dot{z} = \begin{bmatrix} 0 & & 0 & -a_0 \\ 1 & \ddots & & -a_1 \\ & \ddots & 0 & \vdots \\ 0 & & 1 & -a_{n-1} \end{bmatrix} z + \begin{bmatrix} 1 \\ 0 \\ \vdots \\ 0 \end{bmatrix} u \quad \dot{z} = \begin{bmatrix} -a_{n-1} & 1 & & 0 \\ \vdots & 0 & \ddots & \\ & & \ddots & 1 \\ -a_0 & 0 & 0 & 0 \end{bmatrix} z + \begin{bmatrix} 0 \\ \vdots \\ 0 \\ 1 \end{bmatrix} u$$

$$y = cTz$$

Note special structure of  $A$ ,  $b$



# Controllability Form – Proof

$$\dot{z} = (T^{-1}AT)z + (T^{-1}b)u$$

$$T^{-1}T = I \Rightarrow [T^{-1}b \quad T^{-1}Ab \quad \dots \quad T^{-1}A^{n-1}b] = I$$

$$T^{-1}AT = [T^{-1}Ab \quad T^{-1}A^2b \quad \dots \quad T^{-1}A^nb]$$

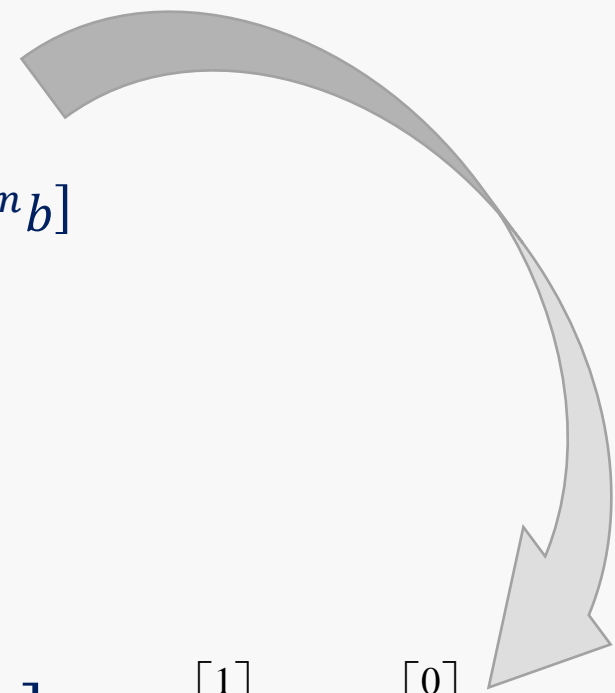
$$T^{-1}b = \begin{bmatrix} 1 \\ 0 \\ \vdots \\ 0 \end{bmatrix} = \begin{bmatrix} 0 & 0 & Y_1 \\ 1 & & Y_2 \\ & 0 & \vdots \\ 0 & 1 & Y_n \end{bmatrix}, Y = T^{-1}A^nb$$

suppose  $\det(\lambda I - A) = \lambda^n + a_{n-1}\lambda^{n-1} + \dots + a_0$ ,

$$\text{C-H Thm} \Rightarrow A^n + a_{n-1}A^{n-1} + \dots + a_0I = 0$$

$$Y = T^{-1}A^nb = -a_{n-1}T^{-1}A^{n-1}b - \dots - a_0T^{-1}b = \begin{bmatrix} -a_0 \\ -a_1 \\ \vdots \\ -a_{n-1} \end{bmatrix}$$

$$T^{-1}b = \begin{bmatrix} 1 \\ 0 \\ \vdots \\ 0 \end{bmatrix}, T^{-1}Ab = \begin{bmatrix} 0 \\ 1 \\ \vdots \\ 0 \end{bmatrix}, \dots, T^{-1}A^{n-1}b = \begin{bmatrix} 0 \\ \vdots \\ 0 \\ 1 \end{bmatrix}$$



# Observability Form

$$\dot{x} = Ax + bu, y = cx$$

$$\mathcal{O} = \begin{bmatrix} c \\ cA \\ \vdots \\ cA^{n-1} \end{bmatrix}, \det \mathcal{O} \neq 0$$

$$T = \mathcal{O} \Rightarrow$$

$$\dot{z} = \begin{bmatrix} 0 & 1 & & 0 \\ & \ddots & \ddots & \\ & & 0 & 1 \\ -a_{n-1} & \cdots & -a_1 & -a_0 \end{bmatrix} z + (T^{-1}b)u, y = [1 \ 0 \ \cdots \ 0]z$$

Note special structure of  $A, c$

# Observability Form – Proof

$$y = cx$$

$$\dot{y} = cAx, \quad cb = 0$$

$$\ddot{y} = cA^2x, \quad cAb = 0$$

$\vdots$

$$y^{(n-1)} = cA^{n-1}x, \quad cA^{n-2}b = 0$$

$$y^{(n)} = cA^n x + u, \quad cA^{n-1}b = 1$$

$$z_1 = cx$$

$$z_2 = cAx$$

$\vdots$

$$z_n = cA^{n-1}x$$

$$z = Sx$$

$$\dot{z}_1 = z_2$$

$$\dot{z}_2 = z_3$$

$\vdots$

$$\dot{z}_{n-1} = z_n$$

$$\dot{z}_n = cA^n S^{-1}z + u$$

$\Rightarrow$

# Observability Form – Proof, cont'd

$$S = \begin{bmatrix} c \\ cA \\ \vdots \\ cA^{n-1} \end{bmatrix}, SS^{-1} = I \Rightarrow \begin{bmatrix} cS^{-1} \\ cAS^{-1} \\ \vdots \\ cA^{n-1}S^{-1} \end{bmatrix} = I$$

$$A^n + a_{n-1}A^{n-1} + \cdots + a_1A + a_0I = 0$$

$$\begin{aligned} cA^n S^{-1} &= -a_{n-1}cA^{n-1}S^{-1} - \cdots - a_1cAS^{-1} - a_0cS^{-1} \\ &= \begin{bmatrix} -a_{n-1} & \cdots & -a_1 & -a_0 \end{bmatrix} \end{aligned}$$

# Summary of SISO Companion Forms

$\boxed{T_1} \quad \dot{z} = \begin{bmatrix} 0 & 0 & -a_0 \\ 1 & \ddots & -a_1 \\ & \ddots & 0 \\ 0 & 1 & -a_{n-1} \end{bmatrix} z + \begin{bmatrix} 1 \\ 0 \\ \vdots \\ 0 \end{bmatrix} u$ $y = \bar{c}z$	$\dot{z} = \begin{bmatrix} 0 & 1 & 0 \\ & \ddots & \ddots \\ & & 0 & 1 \\ -a_{n-1} & \cdots & -a_1 & -a_0 \end{bmatrix} z + \bar{b}u$ $y = [1 \ 0 \ \cdots \ 0]z$
<div style="border: 1px solid black; background-color: #e0e0e0; padding: 5px; display: inline-block;">Controllability</div>	<div style="border: 1px solid black; background-color: #e0e0e0; padding: 5px; display: inline-block;">Observability</div>

$\dot{z} = \begin{bmatrix} 0 & 1 & 0 \\ & \ddots & \ddots \\ & & 0 & 1 \\ -a_0 & -a_1 & \cdots & -a_{n-1} \end{bmatrix} z + \begin{bmatrix} 0 \\ \vdots \\ 0 \\ 1 \end{bmatrix} u$ $y = \bar{c}z$	$\dot{z} = \begin{bmatrix} -a_{n-1} & 1 & 0 \\ \vdots & 0 & \ddots \\ -a_1 & \ddots & 1 \\ -a_0 & 0 & 0 \end{bmatrix} z + \bar{b}u$ $y = [1 \ 0 \ \cdots \ 0]z$
<div style="border: 1px solid black; background-color: #e0e0e0; padding: 5px; display: inline-block;">Controller/phase variable</div>	<div style="border: 1px solid black; background-color: #e0e0e0; padding: 5px; display: inline-block;">Observer</div>

# Transfer Functions from State Space

$$\dot{x} = Ax + Bu$$

$$y = Cx + Du$$

take Laplace transform:

$$sX(s) = AX(s) + BU(s) \Rightarrow [sI - A]X(s) = BU(s)$$

$$Y(s) = CX(s) + DU(s)$$

⇓

$$Y(s) = \left\{ C[sI - A]^{-1} B + D \right\} U(s)$$

⇓

$$G(s) = C[sI - A]^{-1} B + D$$

# Example

$$\dot{x} = \begin{bmatrix} -2 & -2 \\ 0 & -4 \end{bmatrix} x + \begin{bmatrix} 1 \\ 1 \end{bmatrix} u, \quad y = [1 \quad 0] x$$

$$\mathcal{C} = [B \quad AB] = \begin{bmatrix} 1 & -4 \\ 1 & -4 \end{bmatrix} \Rightarrow \text{not controllable}, \quad \mathcal{O} = \begin{bmatrix} C \\ CA \end{bmatrix} = \begin{bmatrix} 1 & 0 \\ -2 & -2 \end{bmatrix} \Rightarrow \text{observable}$$

$$G(s) = C[sI - A]^{-1} B = [1 \quad 0] \frac{\begin{bmatrix} s+4 & -2 \\ 0 & s+2 \end{bmatrix}}{(s+2)(s+4)} \begin{bmatrix} 1 \\ 1 \end{bmatrix}$$
$$= \frac{(s+2)}{(s+2)(s+4)} = \frac{1}{s+4}$$